

# OPTIMIZATION AND SCALE-UP TESTING OF COMPOSITE REPAIR TECHNOLOGIES

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# OVERVIEW

- ◆ Who We Are
- ◆ History
- ◆ Background
- ◆ Repair System and Tests
- ◆ Preliminary Testing (Coupon)
- ◆ Full-scale Testing (Burst and Cyclic)
- ◆ Conclusions

# WHO WE ARE

- ◆ Manufacturer of composite materials designed to comply with ASME, ISO and ACI standards in addition to solutions for protecting directional drill piping
- ◆ State-of-the-art, ISO 9001-certified manufacturing and R&D facility in FL
- ◆ Solution driven products are backed by in-house R&D, engineering, training and site support departments. Also offer accredited continuing education courses

R&D



Manufacturing



Training



# HISTORY

- ◆ China Airlines Flight 611 – Taiwan to Hong Kong
- ◆ February 7, 1980
  - Tail strike on landing
- ◆ May 25, 2002
  - Crash due to metal fatigue and improper repair killing all 225 on board



# BACKGROUND

- ◆ Increased composite material usage leading to more testing programs and awareness
- ◆ Understanding small changes in composite make-up
- ◆ Many testing focused only on “burst testing”
- ◆ This testing was to focus on the effect of the fibers within the repair system on the defect to optimize the two as a system

Anomaly Type / Environment	Current Industry Validation Level of Composite Repair Systems
Corrosion	Well documented
Dents	Well documented
Wrinkle Bends	Documented
Cracks	Limited
High Temperature (up to 100°C / 212°F)	Limited

# FABRIC AND TEST MATRIX

- ◆ Bi-directional E-glass (0/90, stitched) with 2-part epoxy
  - Varying additions of load transfer chopped strand mat
- ◆ Coupon-level
  - Tensile, Flexural, Interlaminar Shear Strength testing
    - Strength, Modulus, and Elongation/Displacement
- ◆ Full Scale
  - Burst test following two five minute pressure holds at MAOP (1,778 psi) and 100% SMYS (2,470 psi)
  - Cyclic tests ranging from 36-72% SMYS (890-1,780psi) at 6 cycles per minute with a target of 275,000 cycles

# FABRIC AND TEST MATRIX

	Ratio of Fiber Hoop:Axial	Load Transfer CSM Addition (oz/yd <sup>2</sup> )	Coupon Level Tests	Full Scale Tests	
Group 1	70:30	0, 4, 8, 12	Tensile	-	-
Group 2	80:20		Flexural	Burst	Cyclic
Group 3	90:10		ILSS	Burst	-

# COUPON TESTING

- ◆ Composite design from industry standards and past testing results used for comparison purposes
- ◆ Tensile modulus and elongation to failure identified as critical elements from design perspective
  - Other elements met by the polymer alone (i.e. temperature limits, viscosity, etc.)
- ◆ This testing was to focus on the effect of the fibers within the repair system and to optimize that piece of the system



# COUPON TEST DISCOVERIES

## ◆ Tensile Testing

- Elongation to failure tended to increase with addition of chopped fiber (*preferable for long-term cyclic testing*)

## ◆ Flexural Testing

- Failure mode of the 0 and 12 oz/yd<sup>2</sup> samples was primarily fiber breakage
- Failure mode of the 4 and 8 oz/yd<sup>2</sup> samples was primarily delamination (*preferred method of failure*)

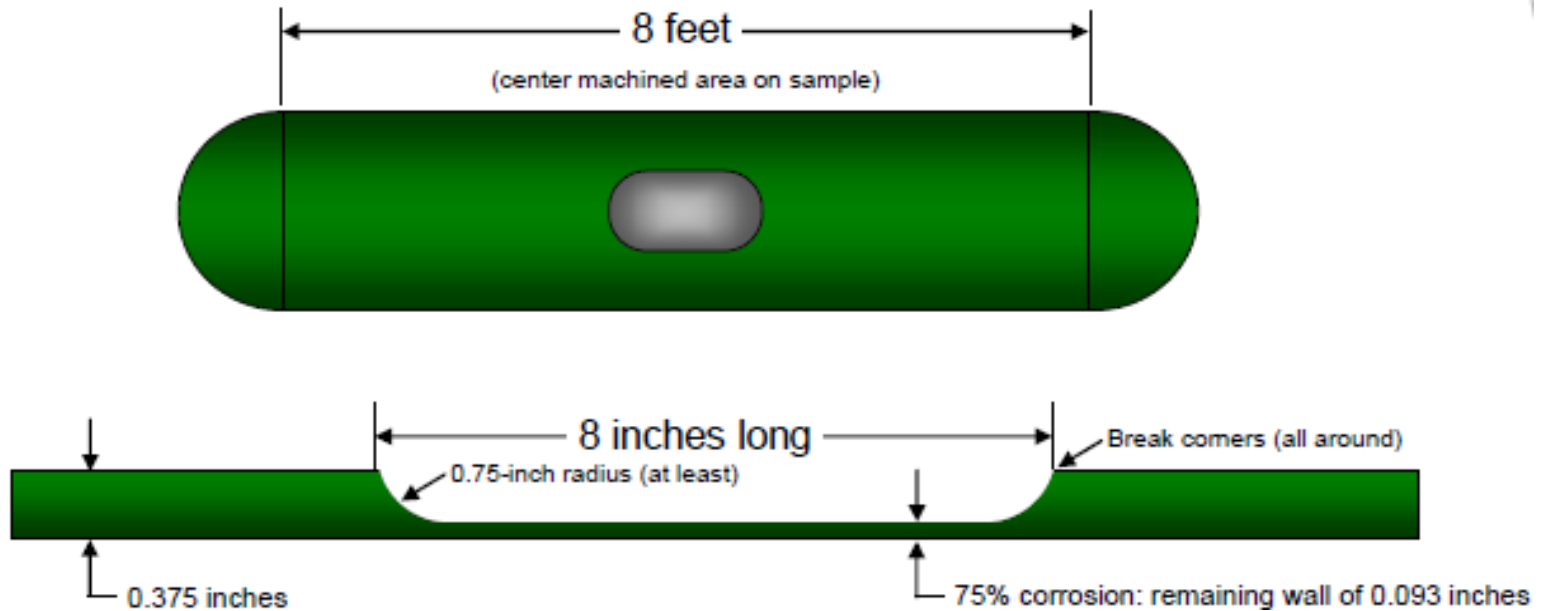
## ◆ ILSS Testing

- Not a significant difference with different percentages of reinforcement

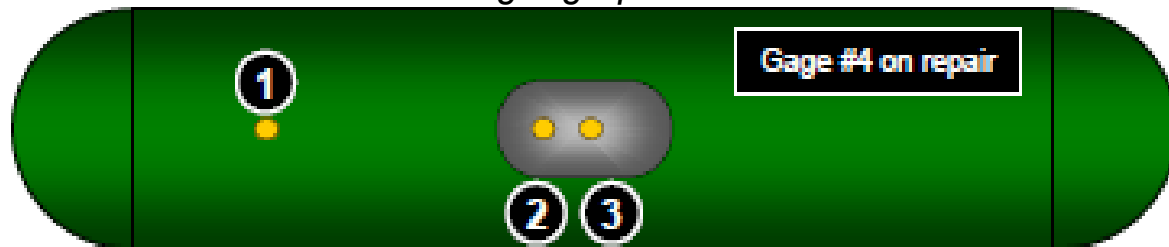
# FULL-SCALE PRESSURE TEST

- ◆ Based on results and discoveries in coupon testing:
  - Test group #1 (70:30) was eliminated entirely
  - Test group #s 2.2, 2.3, 3.2, and 3.3 chosen to continue (80:20 and 90:10 with 4 and 8oz/yd<sup>2</sup> of CSM)
  - Test group #3.1 (90:10 with no CSM) was also chosen but only to serve as a baseline comparison for the other test groups
- ◆ Full-scale pressure test conducted on each sample and biaxial strain data collected
- ◆ Pipe sample and defect created according to drawing using a 12.75-inch x 0.375-in, Grade X52 pipe
- ◆ Strain gauges applied at marked areas

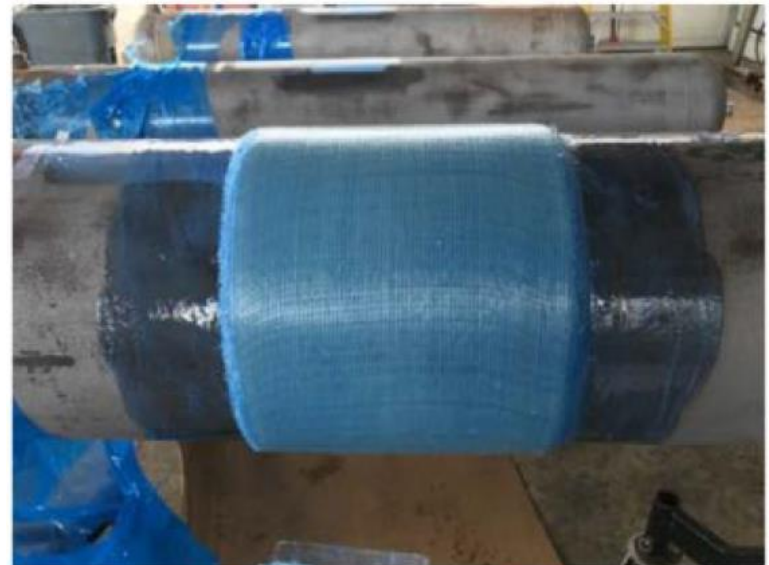
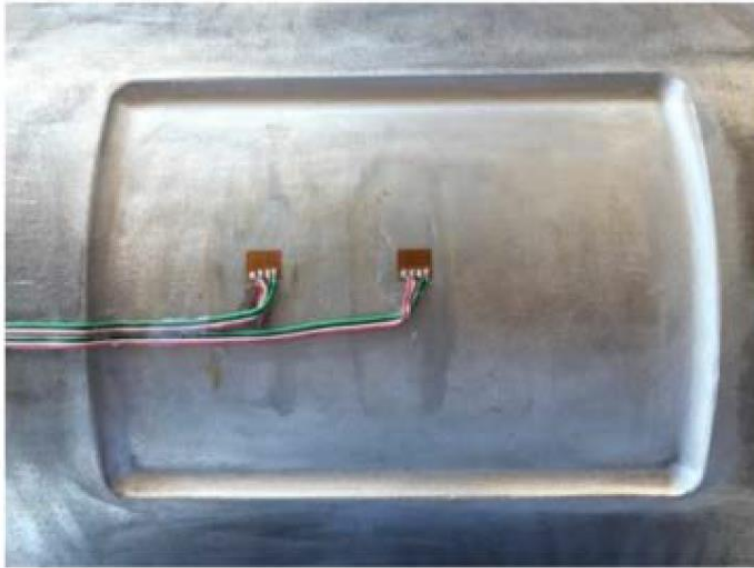
# FULL-SCALE PRESSURE TEST



*Strain gauge placement*



# REPAIR INSTALLATION



# BURST TEST RESULTS

- ◆ Failure pressures noted were all similar in value (*within 4% of the average of all tests*)
- ◆ Nothing conclusive given by the burst pressure burst therefore emphasizing the importance of collecting strain data
- ◆ Determine and rank burst performance for each sample according to strain

# BURST TEST RESULTS

- ◆ For strain data, benchmark targets set based on existing testing from PRCI
  - 3,200 microstrain at MAOP
  - 5,400 microstrain at SMYS
- ◆ Based on the strain readings, all repairs exceeded targets
- ◆ One sample clearly stood out and appears to be the 'best' overall performer

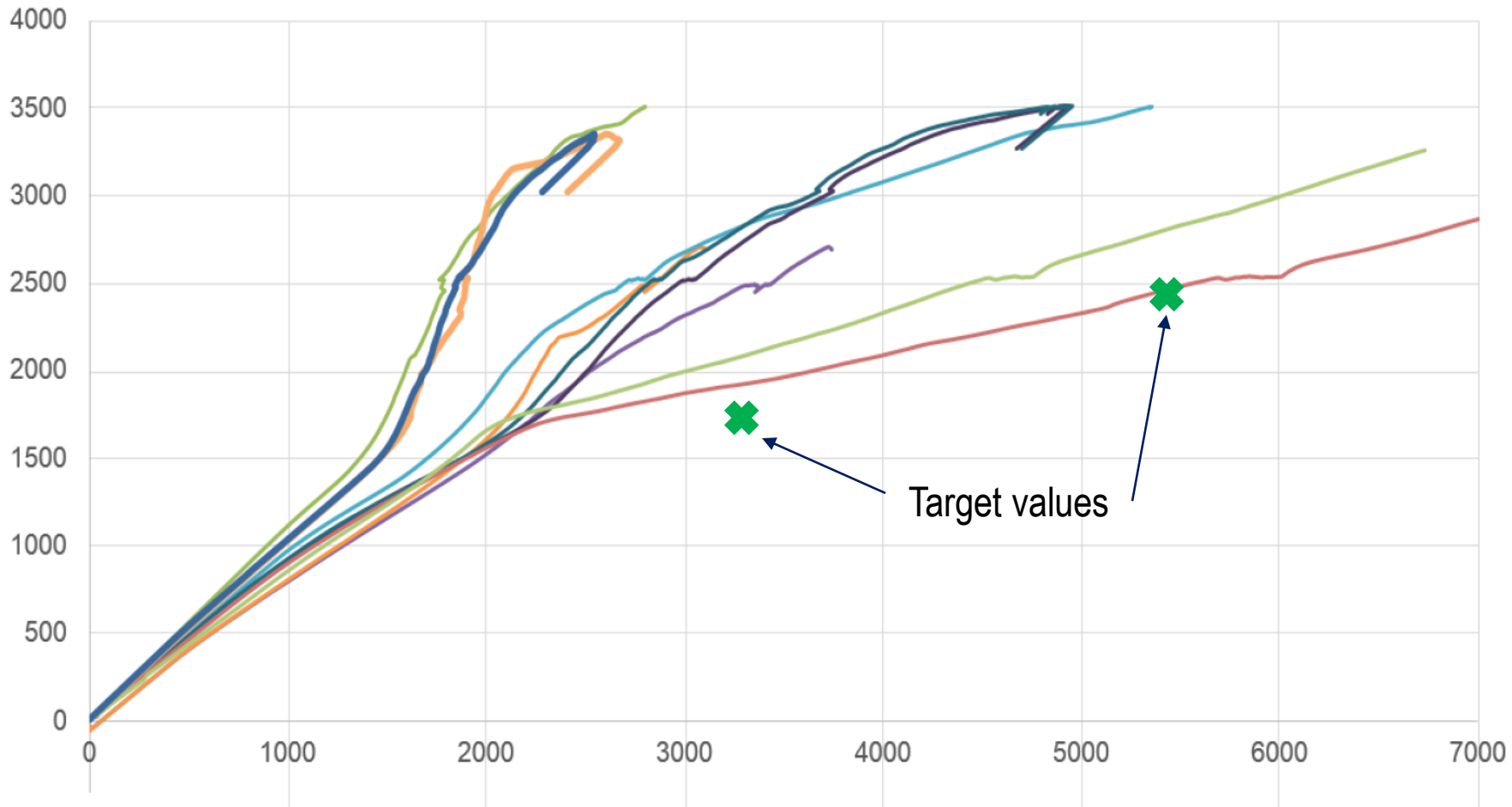
# COUPON AND BURST TESTS

	Tensile Results (average)		
	Strength (ksi)	Modulus (Msi)	Elongation to Failure (%)
Group 1 (70:30)	<b>61.8</b>	<b>3.4</b>	<b>2.2</b>
Group 2 (80:20)	<b>74.7</b>	<b>3.7</b>	<b>2.1</b>
Group 3 (90:10)	<b>66.2</b>	<b>3.6</b>	<b>2.0</b>
	Flexural Results (average)		
	Failure Stress (ksi)	Modulus (Msi)	Displacement (in)
Group 1 (70:30)	<b>78.6</b>	<b>3.1</b>	<b>0.228</b>
Group 2 (80:20)	<b>77.3</b>	<b>3.5</b>	<b>0.193</b>
Group 3 (90:10)	<b>91.0</b>	<b>3.7</b>	<b>0.252</b>
	ILSS Results (average)		
	Failure Stress (ksi)	Modulus (Msi)	Displacement (in)
Group 1 (70:30)	<b>4.7</b>	<b>3.5</b>	<b>0.0380</b>
Group 2 (80:20)	<b>4.7</b>	<b>3.5</b>	<b>0.0390</b>
Group 3 (90:10)	<b>4.8</b>	<b>3.9</b>	<b>0.0365</b>

Burst Results
Pressure (average)
-
<b>4510</b>
<b>4631</b>

Defect yield expected at 765.

# PRESSURE TEST RESULTS





# CYCLIC PRESSURE TEST

- ◆ Based on pressure test results, test group #2.3 (80:20 ratio with 8oz CSM) chosen to undergo cyclic pressure testing
- ◆ Strain measurements taken to identify maximums, minimums, and strain range ( $\Delta\varepsilon$ ) to identify long-term effects on repair system
- ◆ Target a strain range below  $1500\mu\varepsilon$ . Past programs indicate the composite will not degrade significantly due to pressure cyclic fatigue below this range.

# CYCLIC TEST RESULTS

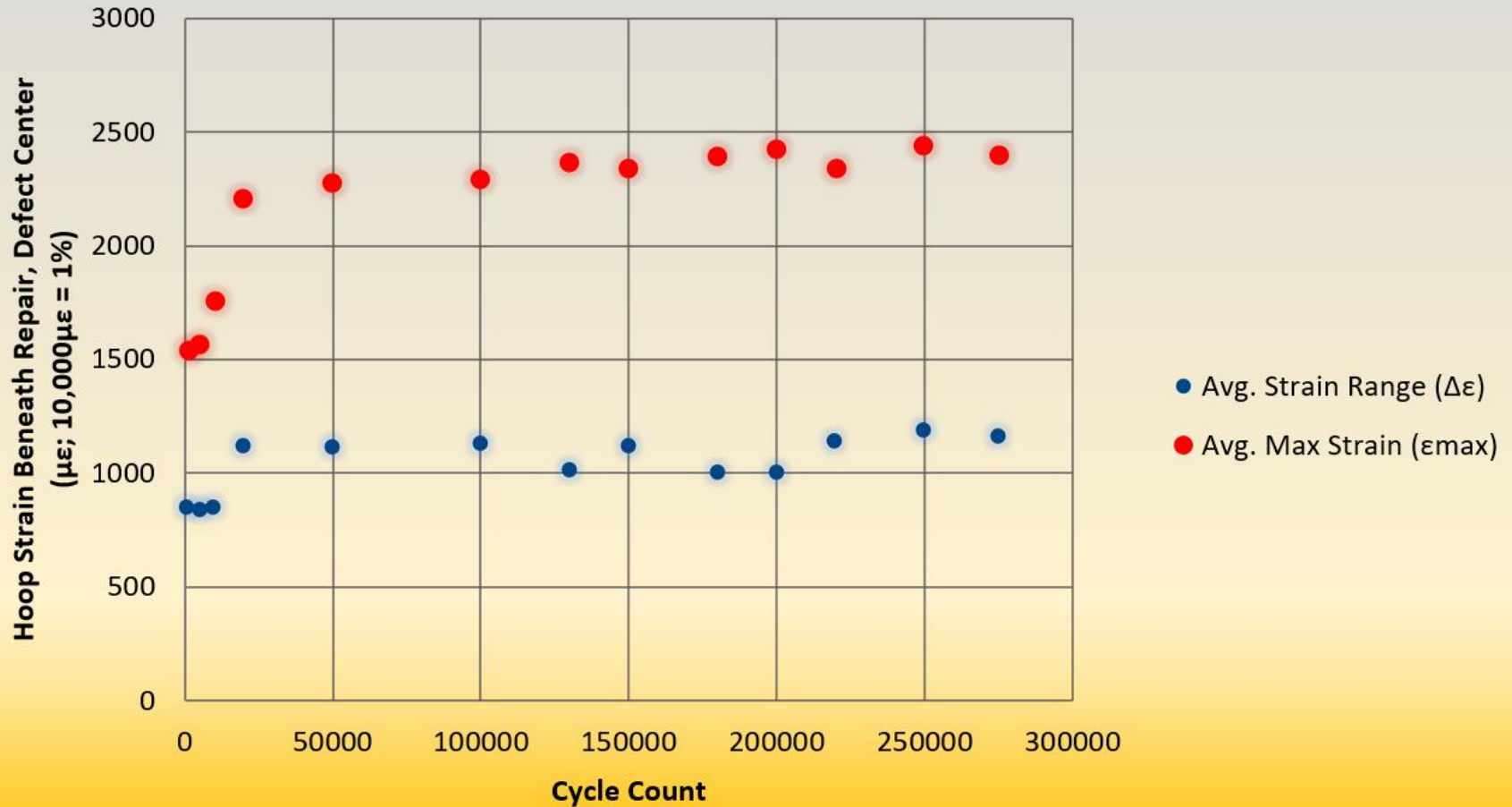
- ◆ All pipe samples completed full “run-out” of 275,000 cycles with no failure or visual damage/wear identified.
- ◆ Relatively constant strain range over long-term usage.
- ◆ Identified early changes, considered “break-in” period of the system while reaching equilibrium of load distribution.
- ◆ Optimized system achieved an average strain range of  $1000\mu\epsilon$  (30% below industry findings) indicating long-term performance will be successful.
  - Target was  $1500\mu\epsilon$  or less.

# CYCLIC TEST RESULTS

Cycle Count	Gauge Location	Hoop Strain Measurements ( $\mu\epsilon$ ) and Strain Range ( $\Delta\epsilon$ )								
		Pipe #14			Pipe #15			Pipe #16		
		Min	Max	$\Delta\epsilon$	Min	Max	$\Delta\epsilon$	Min	Max	$\Delta\epsilon$
100K	Defect center	1424	2404	<b>975</b>	1076	1992	<b>916</b>	1345	2476	<b>1131</b>
	2" off center	2041	3058	<b>1015</b>	1135	1995	<b>860</b>	1416	2391	<b>975</b>
190K	Defect center	1549	2524	<b>980</b>	1148	2053	<b>905</b>	1528	2660	<b>1132</b>
	2" off center	2293	3308	<b>1017</b>	1155	2053	<b>898</b>	1491	2456	<b>965</b>
250K	Defect center	2789	3789	<b>1000</b>	1186	2101	<b>915</b>	1651	2784	<b>1133</b>
	2" off center	3279	4316	<b>1037</b>	1199	2075	<b>876</b>	1548	2524	<b>976</b>
275K	Defect center	<i>Note<sup>1</sup></i>	<i>Note</i>	<b>1051</b>	1150	2046	<b>896</b>	1628	2730	<b>1102</b>
	2" off center	<i>Note</i>	<i>Note</i>	<b>1089</b>	1169	2042	<b>873</b>	1514	2471	<b>957</b>

# CYCLIC TEST RESULTS

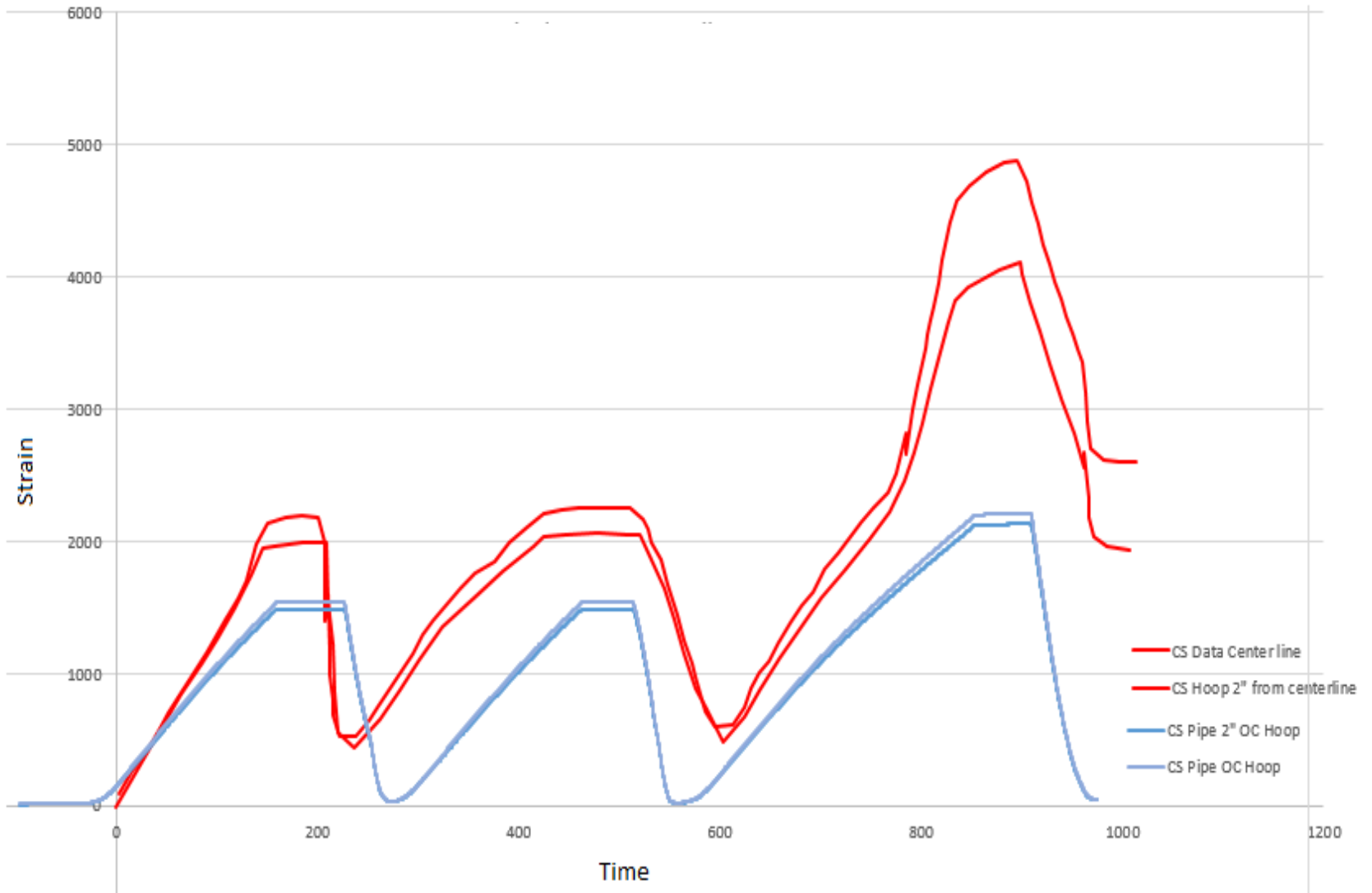
## Strain as a Function of Cycle Count



# FULL-SCALE TEST RESULTS

- ◆ Comparing to identical test on a product by GTI in the 1990's:
- ◆ In burst testing, reduced strain in the defect region by 50% at pressures up to 2000 psi and as pressure increased this reduced strain by over 70%.
- ◆ Also, when the pressure was returned to zero the strain went back to zero showing no permanent yielding in the defect zone.
- ◆ In cyclic testing, reduced max strain in defect region by 50%.
  - Relatively constant strain range over long-term usage

# CYCLIC TEST RESULTS



# CONCLUSIONS

- ◆ Progression from coupon level to full scale testing to optimize a fiber architecture that reduces the strain in severe corrosion defects considering aggressive loading conditions
- ◆ Indications show slight variations can have great impact on the repair system performances
- ◆ Historical data, ASME and ISO standards give baseline requirements but further development is needed to verify applicability of the solution

# WHAT DOES GOOD LOOK LIKE IN 10 YEARS?

- ◆ Solution driven products optimized to address an ever expanding set of deficiencies
- ◆ Legacy, 'work horse' composite repair systems will see reduced roles, some retired
- ◆ Industry wide procedures and protocols developed beyond ASME/ISO
  - Regulators
  - Installers
- ◆ Continued collaboration between manufacturers along with operators and manufacturers



***THANK YOU FOR YOUR TIME!***

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